Analyzing Deep Space Network System Performance and Lessons Learned

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Information about Speaker

- Current responsibility
 - Chief System Engineer of the NASA Deep Space Network (DSN) at Jet Propulsion Laboratory, Pasadena, California
- Area of interests
 - System engineering
 - DSN systems for spacecraft communications and science observations
 - System development
 - Galileo telemetry, antenna arraying, Entry Descent Landing signal detection for Mars Rovers Spirit & Opportunity, Voyager 2 heliopause data return using CSIRO Parkes antenna, Morehead State University's 21m antenna upgrade
 - Consultative Committee for Space Data Systems (CCSDS)
 - Cross Support Transfer Services Area, CCSDS **Engineering Steering Group**
- **Professional recognitions**
 - NASA Exceptional Service Medal, NASA Exceptional Achievement Medal, and several NASA New Technology and **Space Act Awards**
 - **IARIA Fellow**



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Analyzing Deep Space Network System Performance and Lessons Learned

- Nature of deep space communications & navigation
- Key performance metrics to be monitored
- Challenges observed
- Lessons learned



1. Deep Space Communications & Navigation



Deep Space Communications Environment

Long distance communications

- Lunar missions (0.002 AU) to Voyager at 100 AU

	Distance Scale	Power scale
LEO	1	1.0E+00
Moon (e.g., LRO; Human Exploration)	800	6.4E+05
Mars (e.g., Perserverance rover, MRO)	300,000	9.0E+10
Jupiter (e.g. Juno, Europa)	1,560,000	2.4E+12
Saturn (e.g. Cassini)	2,850,000	8.1E+12
Neptune (e.g., Voyager)	8,800,000	7.7E+13
Pluto (e.g., New Horizons)	10,100,000	1.0E+14
Edge of solar system (e.g., Voyager)	30,000,000	9.0E+14

- Low power reception, high power transmission
 - Require large antenna with high G/T and high EIRP
 - Optimal modulation & forward error correction codes
 - Special modes
 - MFSK for Entry Decent Landing when signal is extremely weak
 - Beacon tones for long duration flight to minimize track time and cost
 - Antenna arraying with multiple aperture to increase SNR



Optimized Performance with Variable Data Rate

- Link margin changed by varying elevation during the pass
 - More important at higher operating frequency
 - Adjusting rate per available link margin to maximize data return





Optimized Performance with Error Correction Codes

- Employed special forward error correction codes (e.g., concatenated convolutional/Reed Solomon, Turbo, LDPC) to gain better performance
 - Within 1 dB of AWGN channel capacity
- Minimized system loss in system design
 - Typically less than 1 dB



http://deepspace.jpl.nasa.gov/dsndocs/810-005/208/208A.pdf



Doppler & Ranging Data for Navigation

- Precise measurement of Doppler and Ranging
 - 50 microHz/s Doppler, 1-m ranging
 - 10^{^-14} of 2-30 GHz carrier frequency
- Calibration data needed to minimize systematic errors
 - Earth orientation parameters
 - Media delay in Earth troposphere and ionosphere



2. Key Performance Metrics



Key Performance Metrics

- Some key metrics
 - Signal reception G/T
 - Signal transmission EIRP
 - System SNR loss from antenna to telemetry decoder
- Data collected from operational setting with spacecraft tracking
 - Reflect true operational performance relevant to missions
 - Leverage on abundance of operational data



Sample Display - Categories

DSN Performance Analysis

	Internal PAR Dash Boards	
Dashboards	years year 2021 year 2022	
Linkmargin		
Linkmargin	2010 2017 2018 2019 2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12 Faw	
70m	2016 2017 2018 2019 2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12	
Command Margin	2016 2017 2018 2019 2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12	
Accountability Doppler Accountability	2016 2017 2018 2019 2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12 raw	
Ranging Accountability	2016 2017 2018 2019 2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12 raw	
Delta-DOR Accountability	2016 2017 2018 2019 2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12 raw	
Telemetry Accountability	2016 2017 2018 2019 2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12	
Command Accountability	2016 2017 2018 2019 2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12	
Utilization Ranging Utilization	2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12	1
Command Utilization	2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12	
Antenna Pointing Conscan	2016 2017 2018 2019 2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12 raw quer	ry
Monopulse	2016 2017 2018 2019 2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12 raw quer	y
QQCL Frame Quantity Accountability	2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12	
Frame Quality Accountability	2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12	
Telemetry Latency (Timely)	2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12	
Telemetry Latency (Complete)	2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12	
Frame Gap (Continuity)	2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12	
System Noise Temperature SNT	2016 2017 2018 2019 2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12 raw quer	ry
SNT New Threshold	2016 2017 2018 2019 2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12 810-5	1.1
Radiometric Doppler Noise	2016 2017 2018 2019 2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12 raw	
Ranging Noise	2016 2017 2018 2019 2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12 raw	
Ranging Precal	2016 2017 2018 2019 2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12 raw	
Radio Science Amplitude Stability	2016 2017 2018 2019 2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12 raw	
Allan Deviation	2016 2017 2018 2019 2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12 raw sum	mary
Precal Time Precal Time	2016 2017 2018 2019 2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12 raw	
Signal Acqusition Time Signal Acqusition(nmclog)	2016 2017 2018 2019 2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12 raw	
Signal Acqusition (MIA)	2020 2021 1 2 3 4 5 6 7 8 9 10 11 12 2022 1 2 3 4 5 6 7 8 9 10 11 12	



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Sample Display – System Noise Temperature

	S Band V																																		
DOY	91 1	92 2	93 3	94 4	95 5	96 6	97 7	98 8	99 9	100 10	101 11	102 12	103 13	104 14	105 15	106 16	107 17	108 18	109 19	110 20	111 21	112 22	113 23	114 24	115 25	116 26	117 27	118 28	119 29	120 30	% red	% purple	total pass	DSS	SN dif
14																															43% (6)	57% (8)	14	14	3.9
.5	-		-		-	-				-				-	-				-				-	-		-			-		0%(0)	0%(0)	0	15	515
.6																															0%(0)	0%(0)	0	16	
24																															20%(1)	40%(2)	5	24	14.3
.5																															0%(0)	0%(0)	0	25	
6																															75% (12)	0%(0)	16	26	7.0
84																															38%(3)	25%(2)	8	34	36.8
.7																															0%(0)	0%(0)	0	27	
35																															0%(0)	0%(0)	0	35	
3			•••																												31%(4)	31%(4)	13	43	1.0
6				•	•																										33%(3)	11%(1)	9	36	3.2
15	_		_										_							_									_		0%(0)	0%(0)	0	45	
54	•		-						-				•							•									-		50% (3)	50% (3)	6	54	1.0
5	_		_	_	_	_		_			_	_	_		-		_							_	_						0%(0)	0%(0)	0	55	
5							_	-																							48%(11)	43%(10)	23	63	2.9
50						•	-	-							•	•															38%(3)	25%(2)	8	56	0.6
12														•																	11%(1)	89%(8)	9	65	-7.7

Legend SNT Delta (K) >SNT cd=90 +2 < SNT cd=0 -2 SNT cd=90 > SNT cd=0 Fix Value/bad SNT



Sample Display – Link Margin

									DS	SN Tel	emetr	y Link	Mar	gin Pe	rform	ance Ana	lysis Da	ashboa	ard A	pr <mark>20</mark> 2	22 hel	p										
DOY	91	92	93	94	95	96	97	98	99	100	101	102	103	104	105	106	107	108	109	110	111	112	2 113	114	115	5116	117	118	119	120	Average	
SCID-BAND	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	linkmargir	טוז
ACE-S 92																															19.52	ACE-S
CHDR-S 151																															13.03	CHDR-S
DART-X 135																															12.69	DART-X
DSCO-S 78																															19.24	DSCO-S
EMM-X 62																															10.77	EMM-X
GTL-S 1																															20.57	GTL-S
JNO-X 61																															05.12	JNO-X
JWST-K 170																															12.62	JWST-K
JWST-S 170																															13.81	<u>JWST-S</u>
LRO-S 85																															19.00	LRO-S
LUCY-X 49																															05.29	LUCY-X
M01O-X 53																															02.54	<u>M010-X</u>
MEX-X 41																															08.12	MEX-X
M20-X 168																															15.22	<u>M20-X</u>
MMS1-S 108																															11.30	MMS1-S
MMS2-S 109																															11.08	MMS2-S
MMS3-S 110																															10.83	MMS3-S
MMS4-S 113																															10.89	MMS4-S
MRO-X 74																															05.26	MRO-X
MSL-X 76																															13.72	MSL-X
MVN-X 202																															11.34	MVN-X
NHPC-X 98																															04.58	NHPC-X
NSYT-X 189																															07.84	NSYT-X
ORX-X 64																															10.15	ORX-X
PLC-X 5																•															04.03	PLC-X
SOHO-S 21																															11.09	SOHO-S
SPP-K 96																•															03.57	<u>SPP-K</u>
STEREOA-X 234																•															12.71	STEREOA-X
TESS-K 95																															-5.06	TESS-K
TESS-S 95																•															11.52	TESS-S
THB-S 192																															09.79	THB-S
THC-S 193																•															09.94	THC-S
TRACE-X 143																															05.72	TRACE-X
VGR1-X 31																															04.34	VGR1-X
VGR2-X 32																															04.86	VGR2-X
WIND-S 8																															07.21	WIND-S
XMM-S 60																															09.32	XMM-S
DOM	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30		

Legend Link Margin (dB) ■ < 0.0 0.0 < ■ < 2.0 2.0 ■ Array ■ ■ (tall) shadow



3. Challenges Observed



Challenges with Antenna Gain Measurements

- Antenna Gain (G) is a key parameter of system G/T
 - Affected by
 - antenna beam illumination in its optical path
 - · antenna main reflector surface accuracy
 - antenna surface deformation during the track, especially at low elevations
 - Determinable only with special calibration with known radio source
 - At the time of antenna commissioning or special need for calibration
 - For spacecraft tracking, assume antenna performed at level determined by calibration



Figure 15. DSS-26 (Goldstone) X-Band Receive Gain versus Elevation Angle, X/Ka-Mode, 8420 MHz





Figure 25. DSS-26 (Goldstone) Ka-Band Receive Gain versus Elevation Angle, X/Ka-Mode, 32000 MHz



Challenges with System Noise Temperature Measurements

- System noise temperature (SNT) is part of the G/T
 - Measurements done with and without added calibrated noise enables determination of system noise temperature (Y-factor method)
 - Typically measured during the spacecraft tracking passes
 - Varied by elevations
 - Affected by weather
 - Measurements have many exceptions that need to be accounted for in the processing algorithm
 - Frequency and LNA device/path dependent
 - Algorithm tailored to low SNR, not accurate for very strong signal
 - Noise contribution from planetary bodies within the antenna beam, e.g., Moon
 - Operator override, e.g., disabling of SNT measurements
 - Uncalibrated noise diode, e.g., overdue calibration
 - Other anomalous behaviors (shown in next chart)



Sample of Nominal SNT





- Similar profile between actual and predicted SNT
- Averaged Delta SNT between actual and predicted < 5 K
- Smooth curve. No spike or large offset

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Sample of Anomalous SNT



Large variations in mid track







- Significant delta SNT
- Oscillating profile, presence of large spikes

• A simple threshold method would not sufficiently characterize the SNT measurements



Challenges with EIRP Measurements

- Signal transmission EIRP is partly inferred
 - EIRP = Transmitting Power * Antenna Gain
 - Antenna gain assumed, not measurable
 - Transmitting power is monitored
 - But measurement point is before the radiation end point of the ground antenna



Challenges with Link Margin Characterization

- Link Margin = Operating SNR Decoder threshold
 - Objective: assess robustness of the link on data transmission/reception
 - Affected by:
 - antenna pointing
 - accuracy of modelled EIRP & G/T parameters, for both flight and ground equipment
 - decoder threshold offered by type of error correcting codes and code rate
 - system loss contribution from analog-to-digital converter, filters, cabling, etc.
 - weather conditions
- Operating SNR
 - SNR estimate is noisy, within few tenths of dB
- Decoder threshold
 - Code information data not as reliably accurate
- Observation Spacecraft with dual frequencies operates with different margin for each frequency link



4. Lessons Learned



Lessons Learned

- Imperfect characterization of system performance
 - Measurements could be closed to truth, but require some extrapolation/assumptions, such as assumed antenna gain
- Operations data from spacecraft tracking vs. specific test data at commission
 - Operations data tend to be different from calibration measurements that are done by experts at time of system commission
 - With much larger variation and not necessarily conform with nominal behaviors
 - Useful for relative performance comparison between antennas
 - Harder to validate against the design specifications
- Data visualization
 - Need to account for configuration variations
 - Subgroup data in common configuration for correct interpretation, e.g.
 - SNT received only, transmitted & received, frequency-specific, high vs. low SNR, exclusion of planetary background cases
 - Link Margin frequency specific, antenna size



Lessons Learned

- Algorithm adaptive
 - Need to adapt to the fact that the system under measured changes over time
 - Need to adapt to changes in input data as the system under measured evolves over time
 - e.g., specific events in monitor & control log, changes in performance model
 - Need to check on additional data that affect validity of measurements
 - e.g., validity of noise diode calibration
- Subject matter expertise and neural net algorithm
 - Subject matter expertise required for proper accounting of system behaviors
 - Neural net algorithm could be helpful, e.g., data classification of SNT behaviors between nominal and anomalous categories



Acronyms

- AU Astronomical unit (150,000, 000 km)
- AWGN Additive White Gaussian Noise
- dB decibel
- DSN (NASA) Deep Space Network
- EIRP Equivalent Isotropic Radiating Power
- G/T Gain/Temperature
- K Kelvins degree
- LDPC Low Density Parity Check
- MFSK Multiple Frequency Shift Key
- RF Radio Frequency
- SNR Signal to Noise Ratio
- SNT System Noise Temperature

